

ECATS SARC conference 2020-10-14



Harmonic Forcing From Distortion in a Boundary Layer Ingesting Fan

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F100 Whole aircraft CFD results used for realistic BC

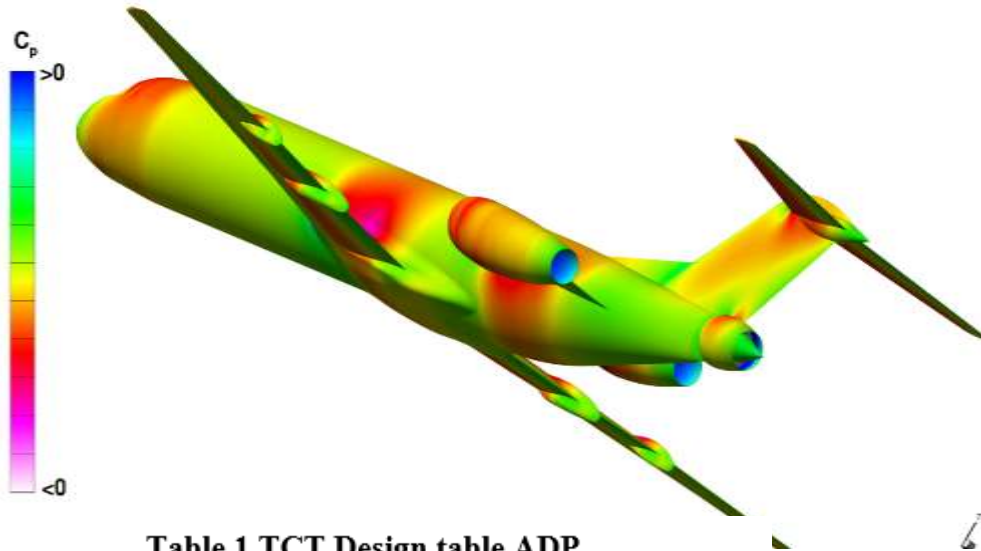
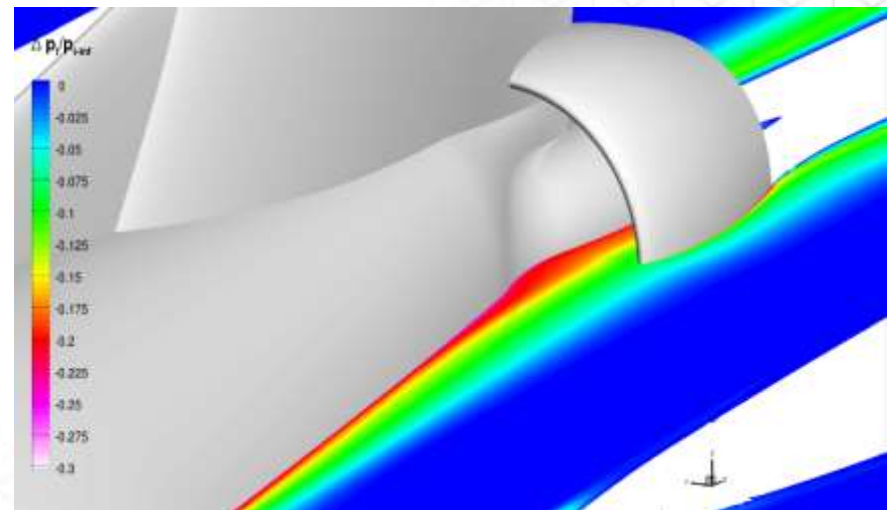
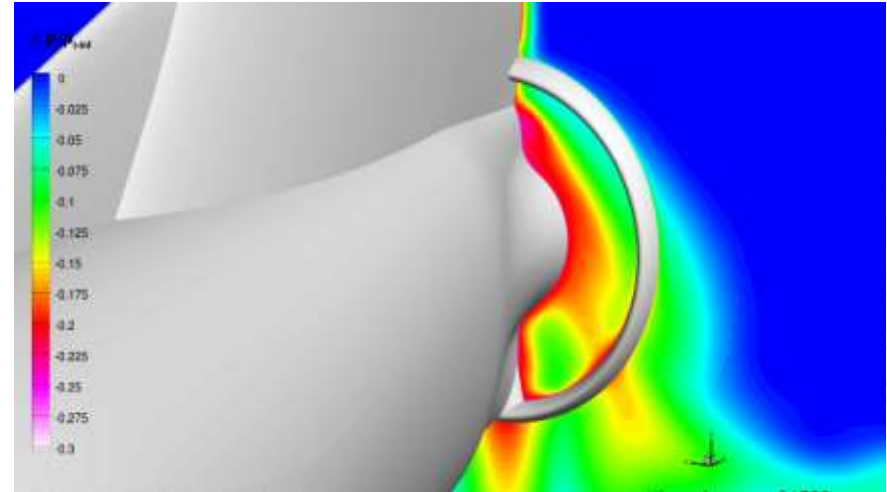


Table 1 TCT Design table ADP

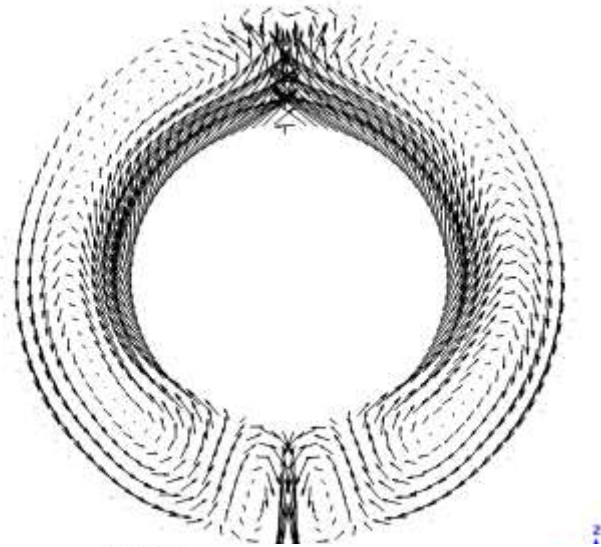
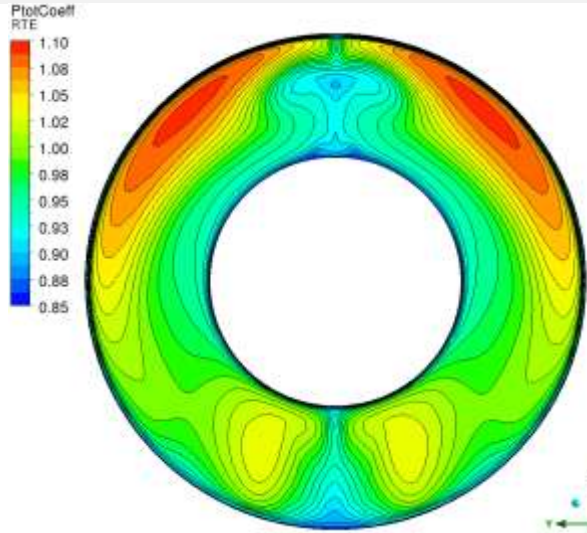
Diameter	[m]	1.2
Shaft speed	[rpm]	4419
PRTT	[-]	1.3
Fan efficiency	[-]	90%
Electric power target	[kW]	1000
Altitude	[ft]	35 000
Flight speed, Ma	[-]	0.74

Performance presented in GT2020-2487, with M.Laban, NLR



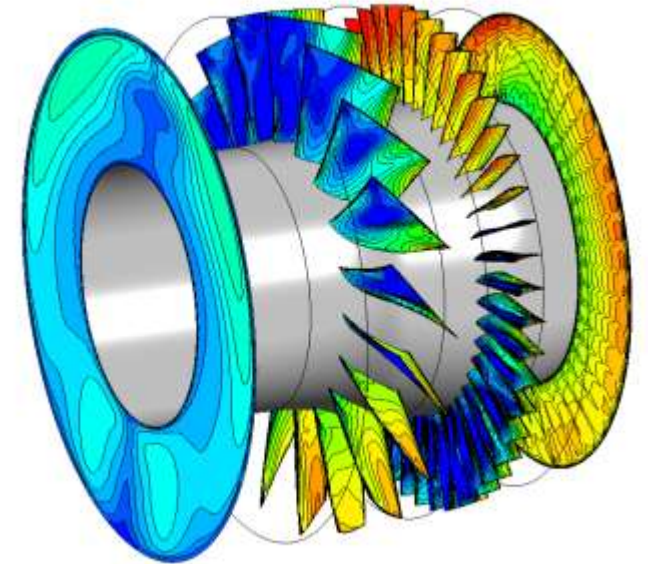
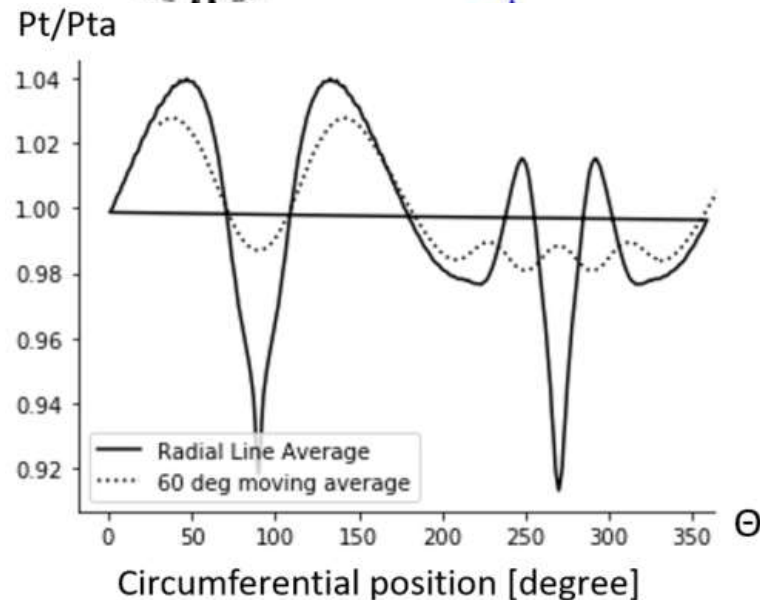
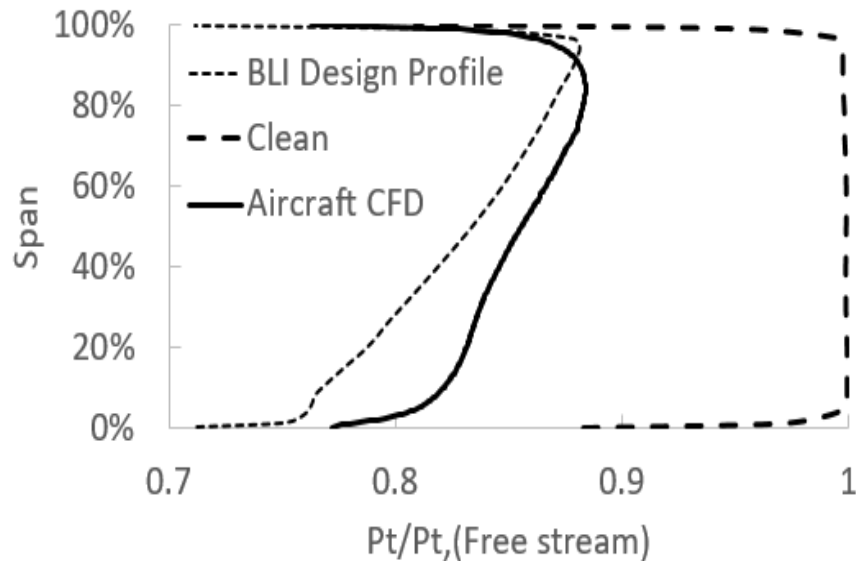


Computation set-up for the fan, 360 unsteady model in CFX



Inlet field Total pressure and swirl velocity vectors mapped from whole aircraft CFD

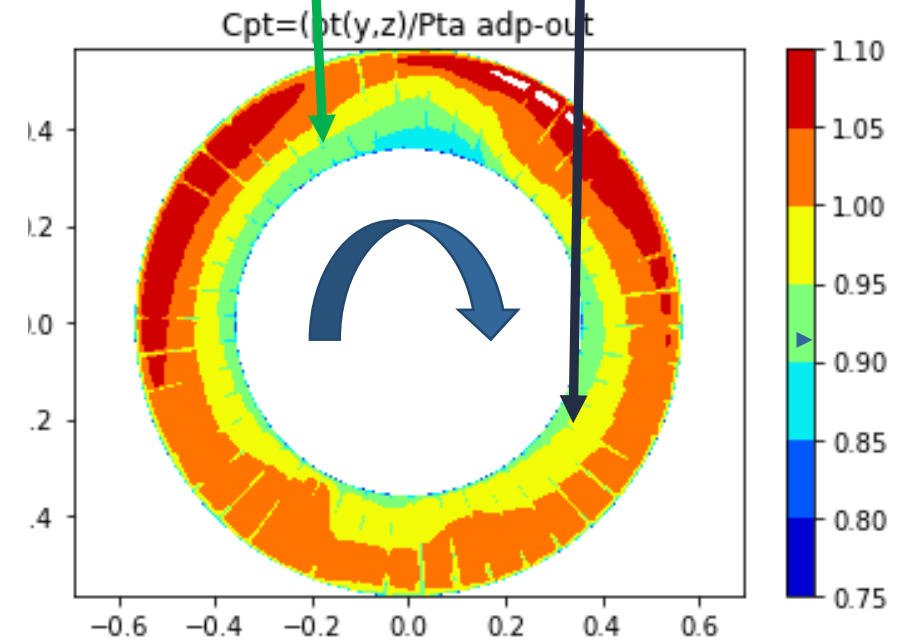
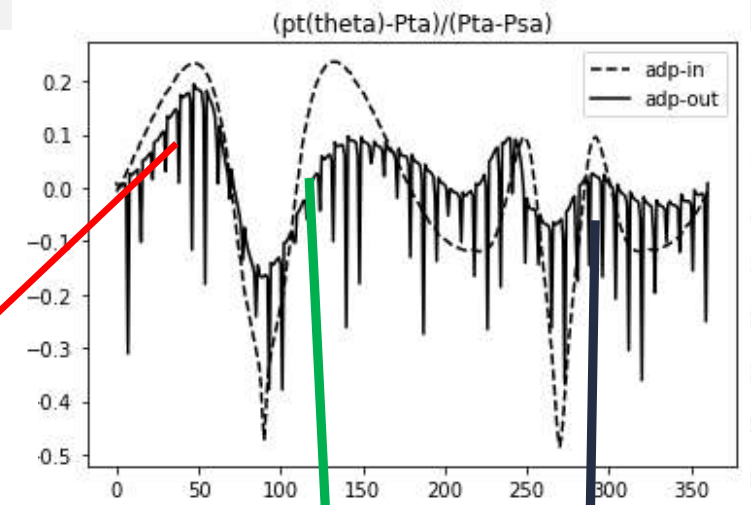
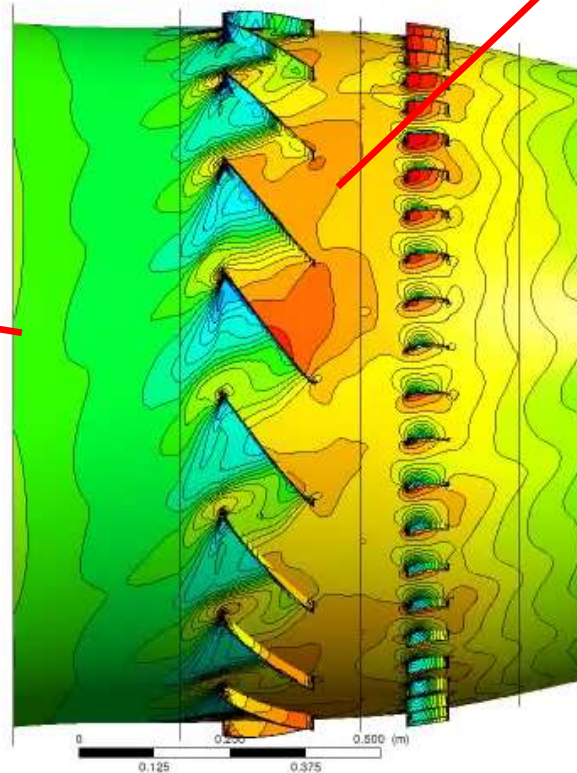
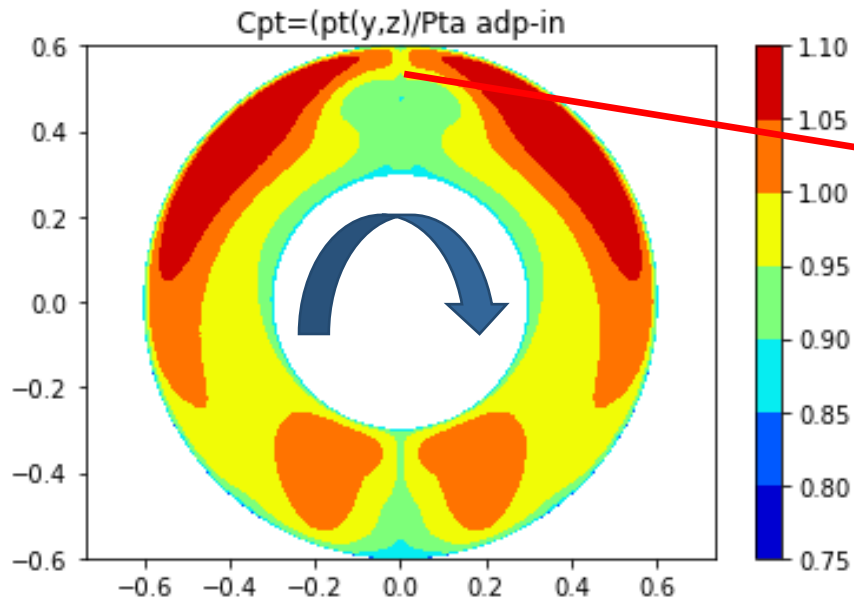
$$DC_{60} = \frac{P_{t,60} - P_{t,ave}}{P_{t,ave} - P} = 12\%$$





Attenuation and unsteady pressure field

Distortion is **attenuated** by the blade load increase coming through the low P0 region

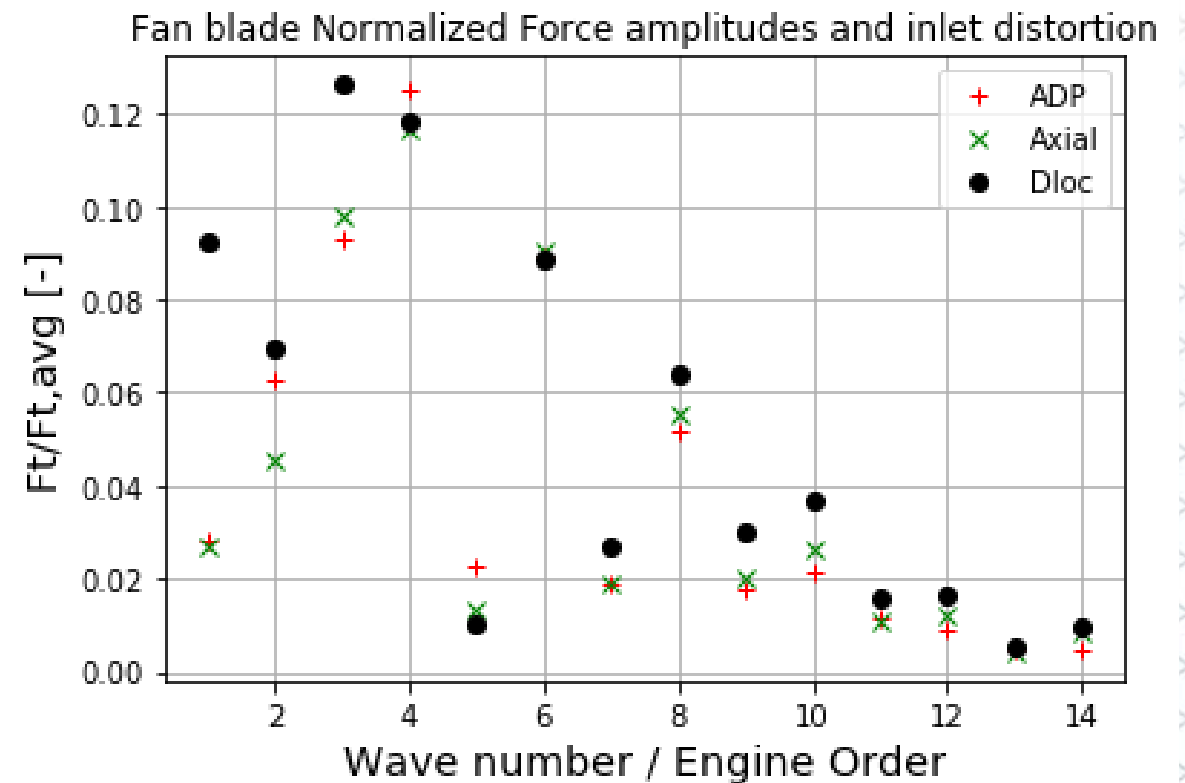
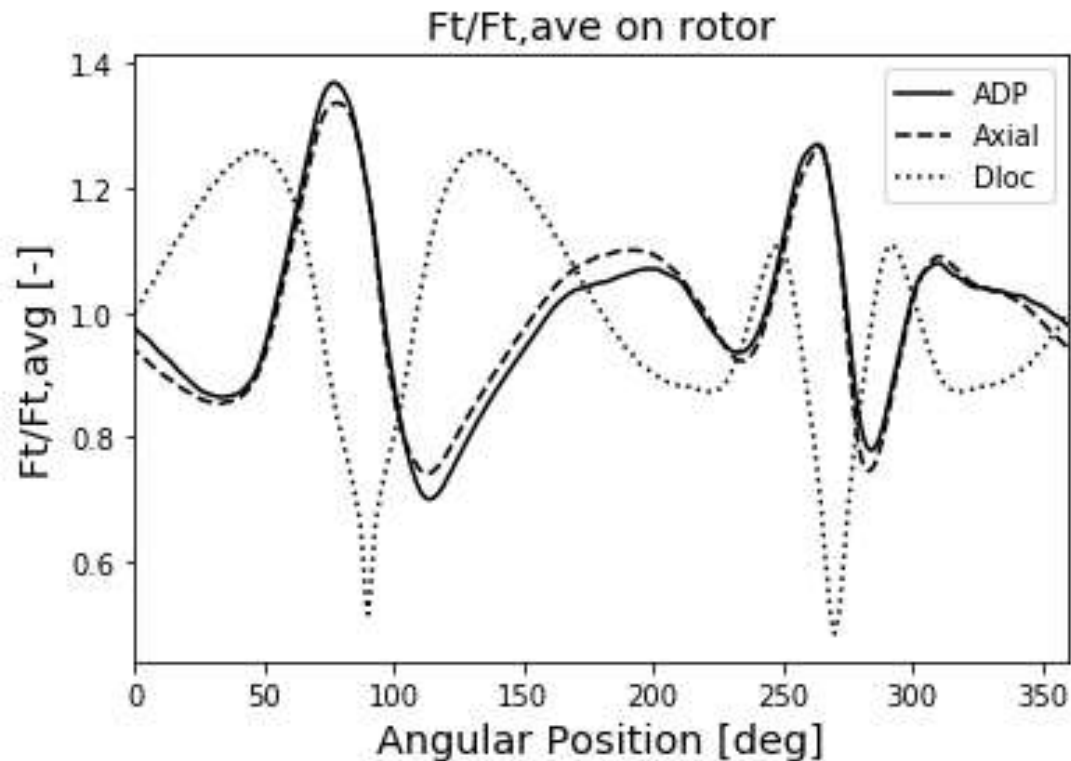


Forcing and distortion correlation at ADP

Good **correlation** with Fourier components of P0 variation, but a **large force amplitude**

The swirl velocity components contributes **marginally**

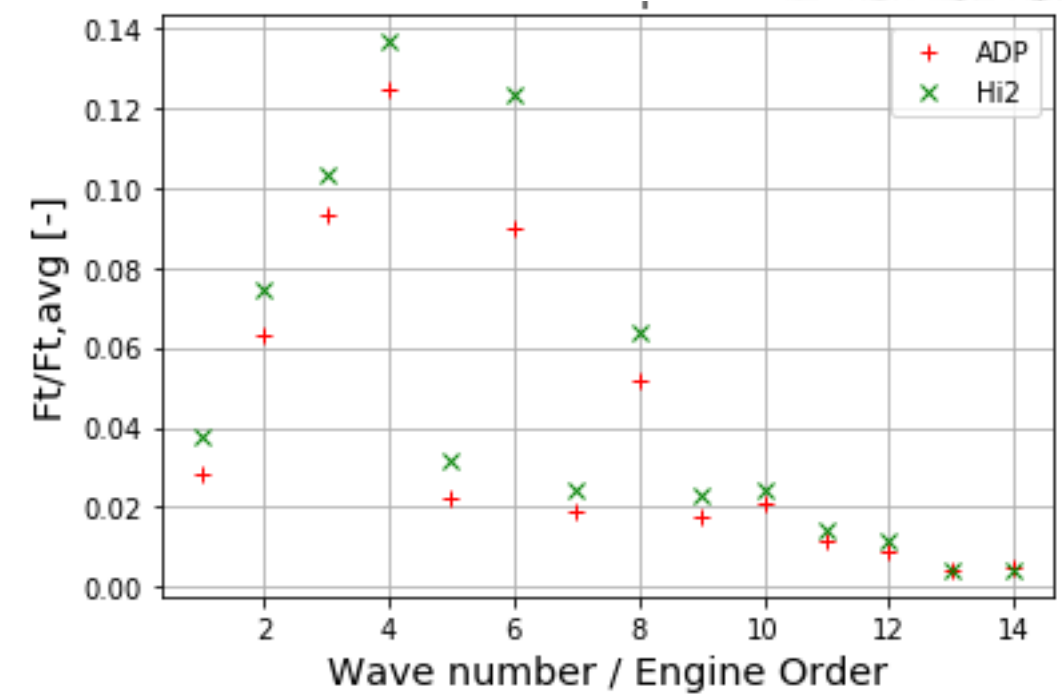
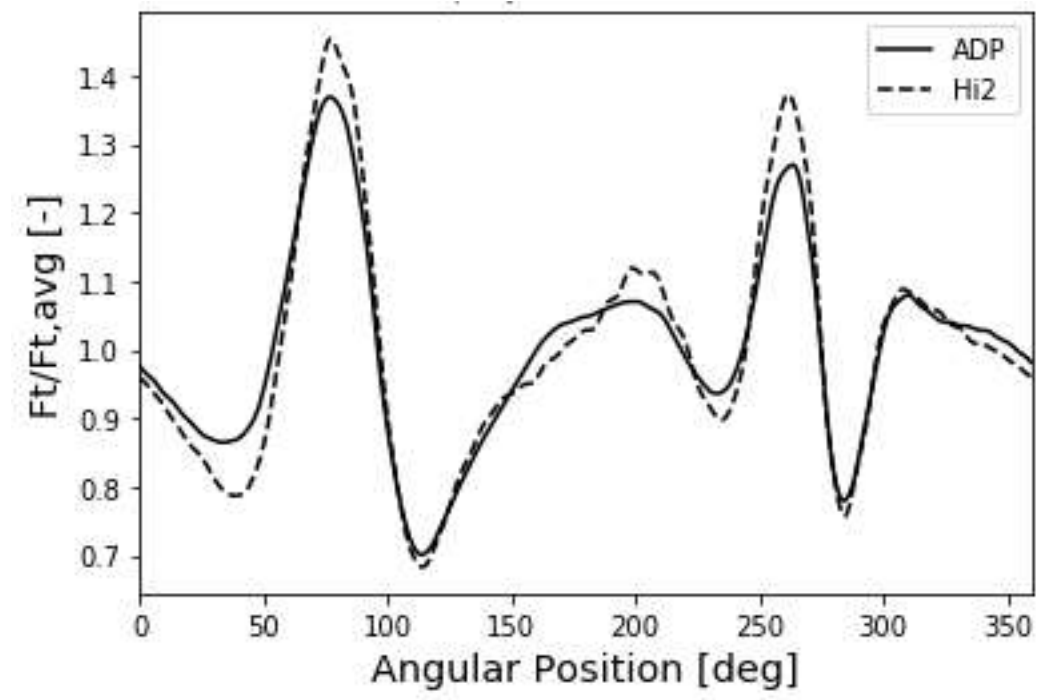
At 10 EO the forcing has dropped off, odd EO earlier



Off-design effect on forcing

Reducing the mass flow 91 % of ADP with the inlet BC

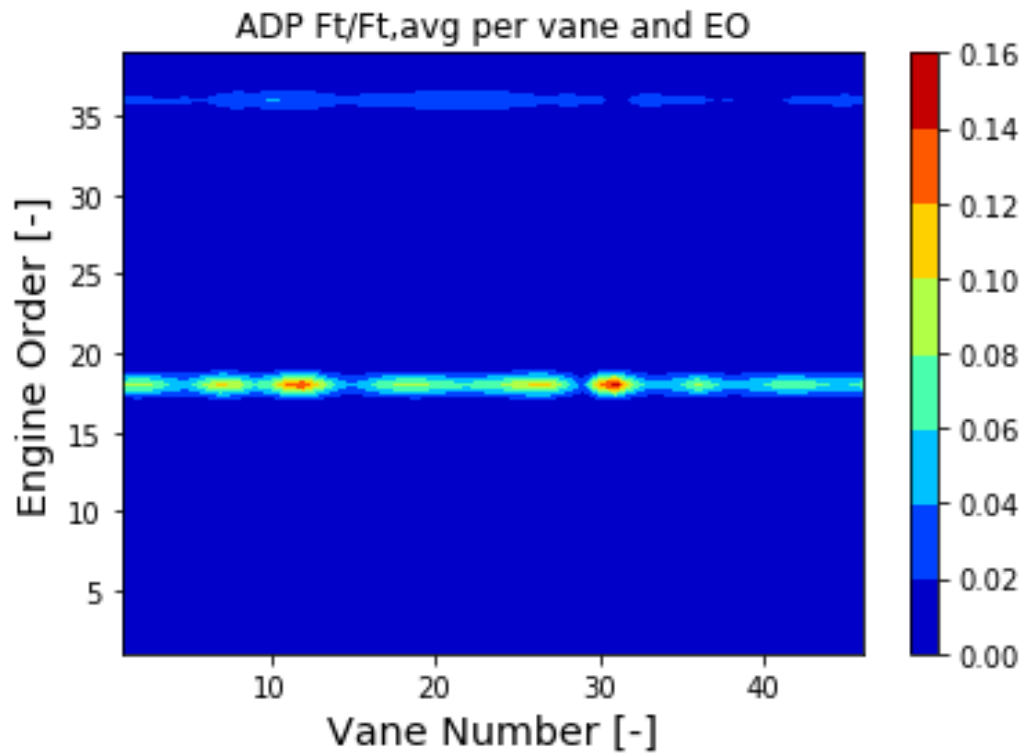
- **increases** normalized forcing **moderately**



Vane harmonic response

Vane responses at 18/rev, 36/rev vary with **position** and **load**

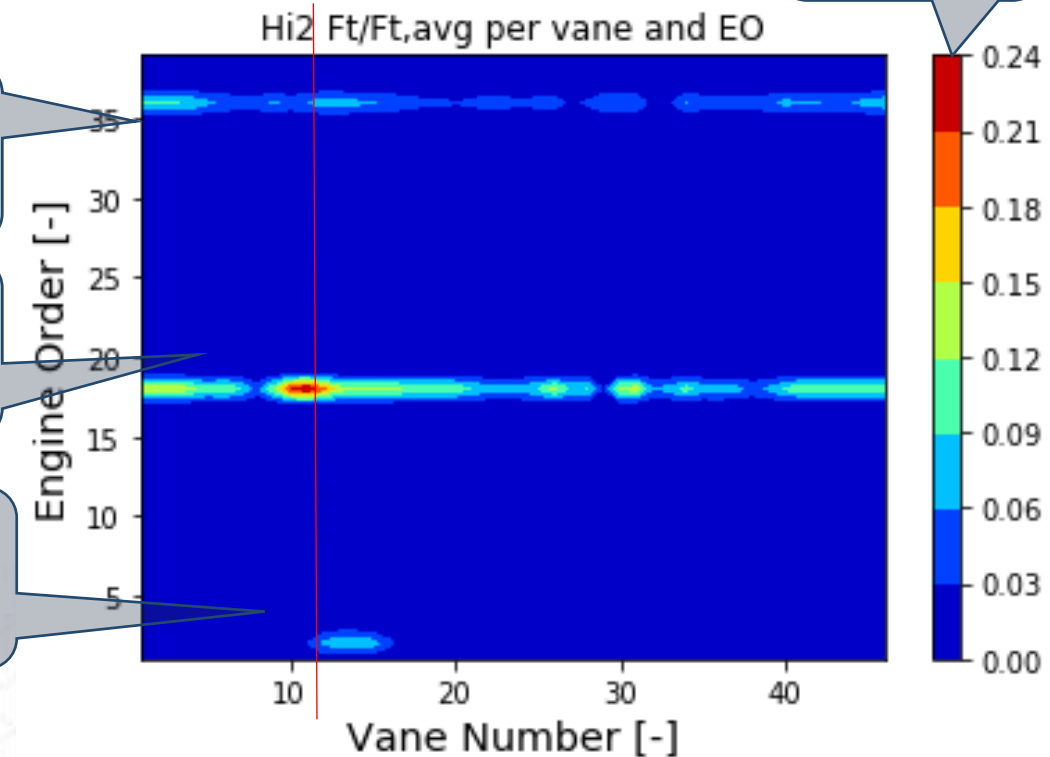
Some low EO response occurs at the higher load



More
36/rev

Changes
18/rev

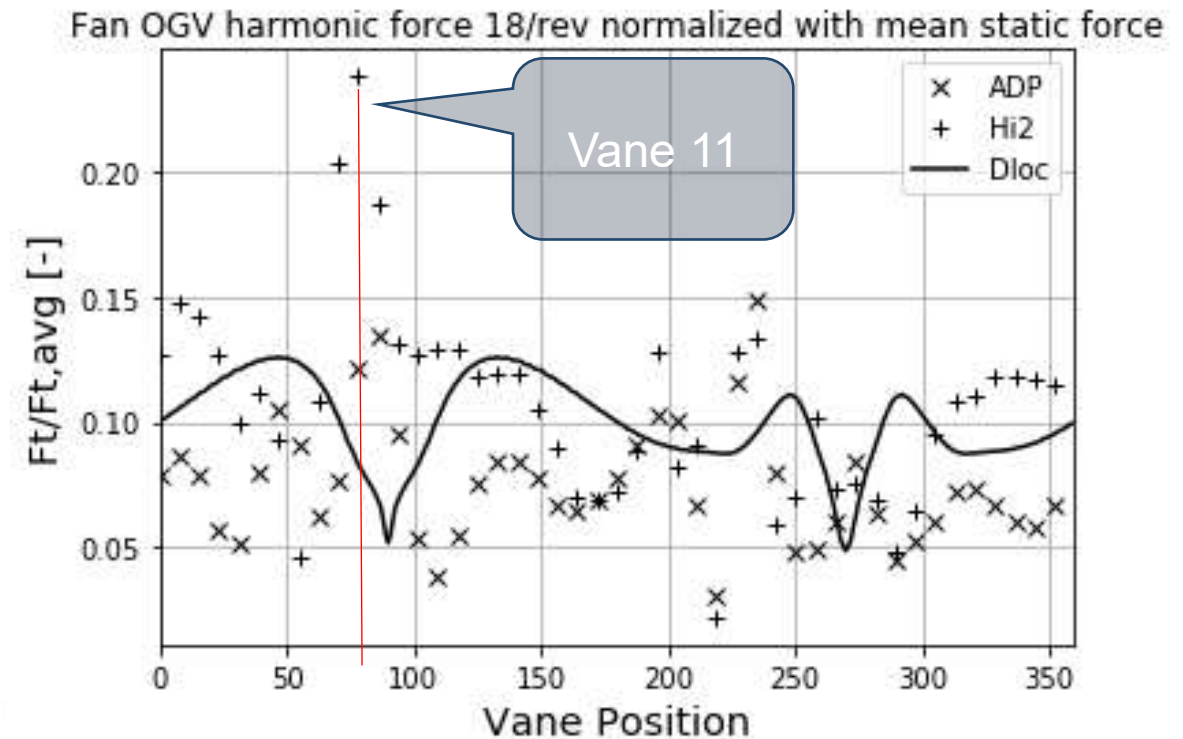
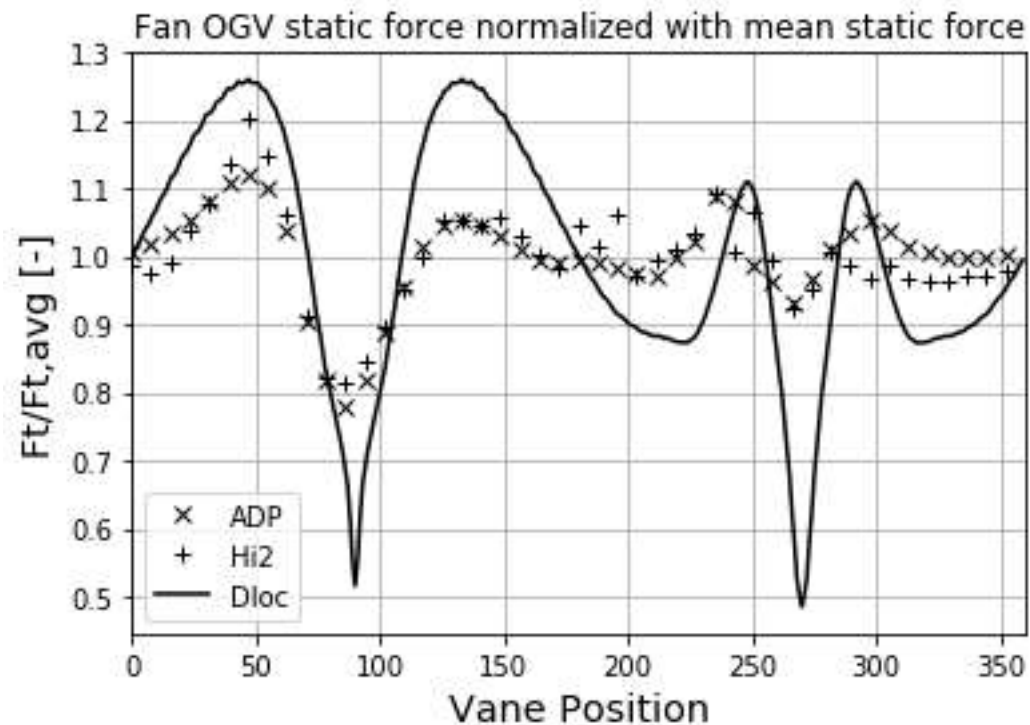
Subharmonic
~1EO



Vane response

Static forces are clearly correlated to the distortion pattern

At a lowered mass flow vanes in the highly loaded sector shows higher response



Summary and Conclusion

In this case – Pt distortion dominates, swirl effects are small

Rotor harmonic forcing correlates well with the distortion parameter for most wavenumbers.

A larger variation of loading on the stator is observed

Results suggest that blades aerodynamics copes with this level of distortion, but also that improvements may be made to the design