



IMPROVEMENT OF AIRPORT LOCAL AIR QUALITY ASSESSMENT

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OUTLINE



- Motivation
- Emission Inventory
- Methods and Materials
- Results
- Concluding remarks

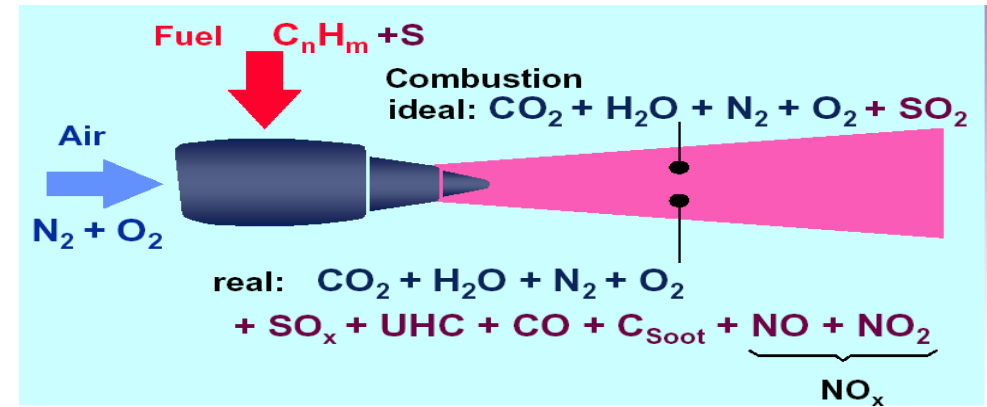
MOTIVATION



Airport air quality is crucial issue



Kyiv International Airport layout (part of the city)



- ✓ increasing air traffic trend;
- ✓ rising tensions of airports expansion ;
- ✓ growing public concern with air quality around the airport (adverse impact on human health of NO_x and PM_{10}).

the development of civil airports should take into account environmental restrictions on the growth of air traffic.

MOTIVATION



Airport Emission Inventory

→ Sources of Airport pollution we need to consider :

Aircraft – departs/Arrivals

- Aircraft Take-off/Landing/Taxi/Queues
- Engine start-up
- APU/GPU



Ground Service Equipment

- Special equipment



Vehicles

- Roadways (landside et airside)
- Parking lots



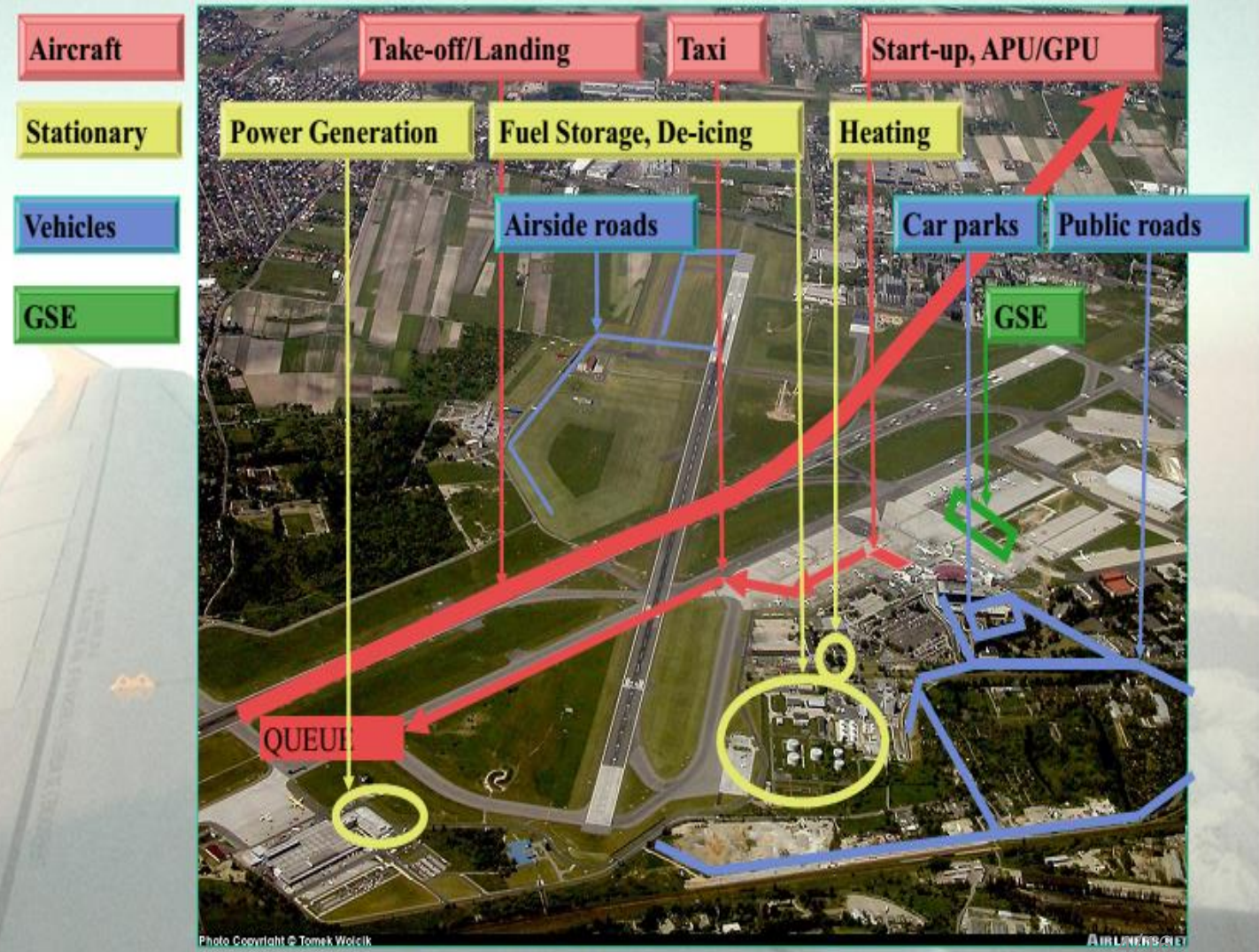
Stationary Point Sources

- De-icing equipment
- Fuel storage
- Heating plant
- Generating stations

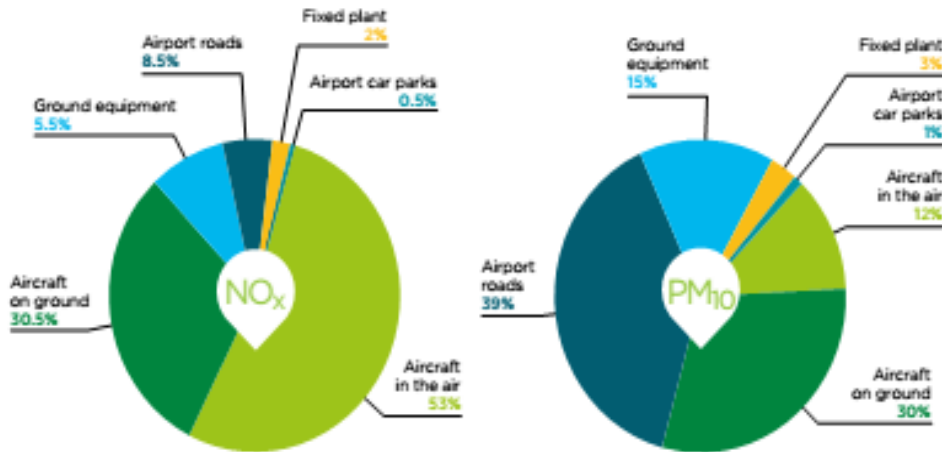


Stationary Area Sources

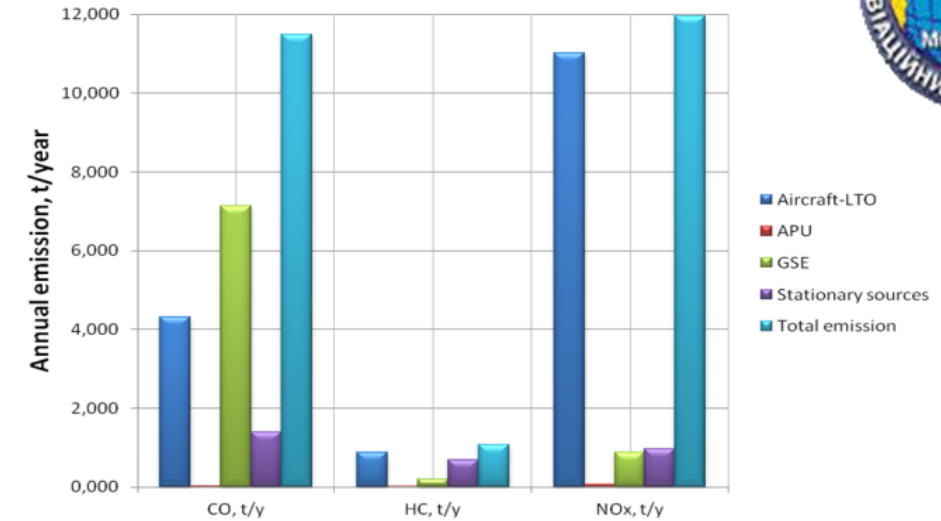
- Fire Training
- Grass cutting
- Fuel storage



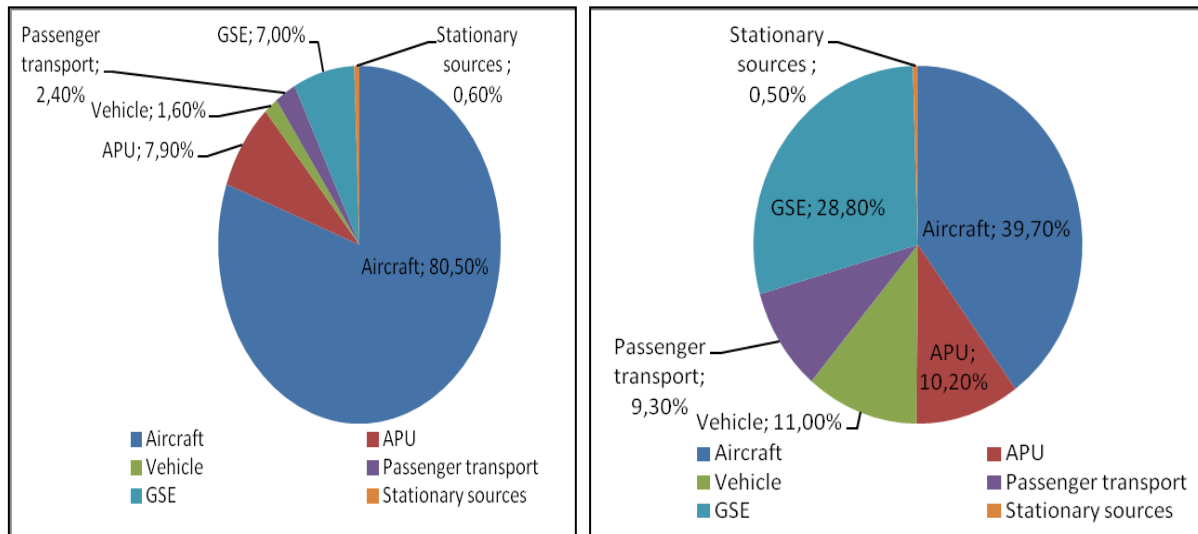
EMISSION INVENTORY



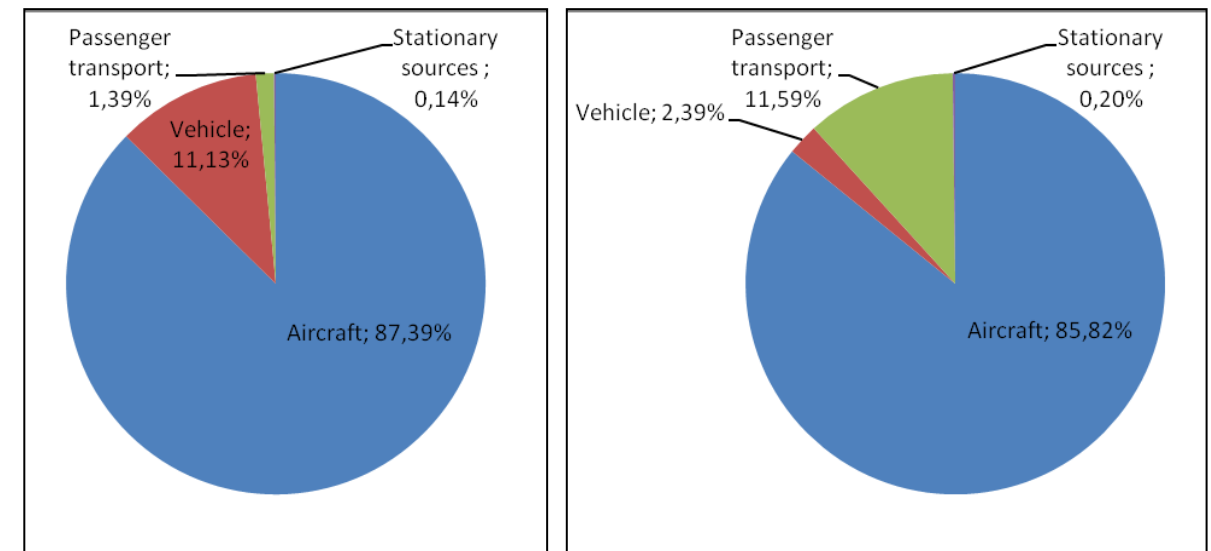
NO_x (a, for Gatwick) and PM₁₀ (b, for Heathrow) emissions



The total annual emission from the sources at Dnipropetrovsk Airport



NO_x (a) and PM₁₀ (b) emissions at Frankfurt Airport



NO_x (a) and PM₁₀ (b) emissions at Boryspil Airport

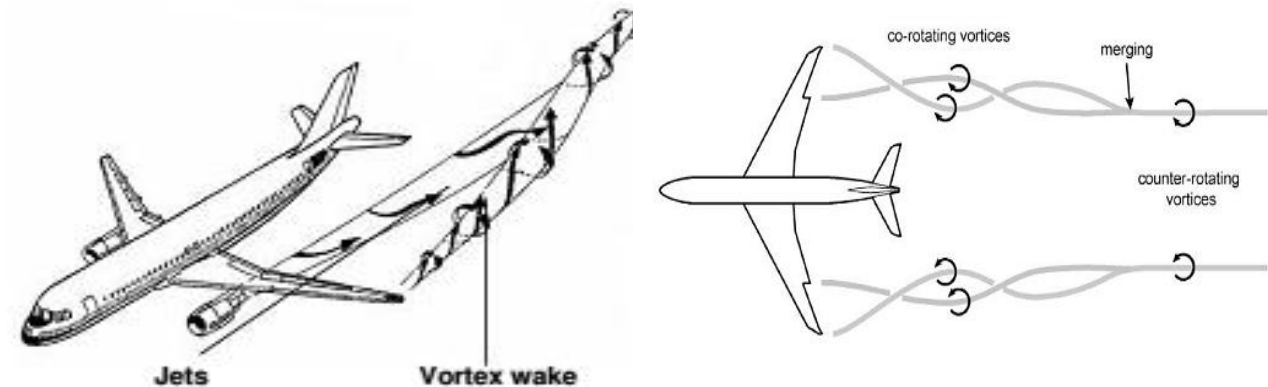
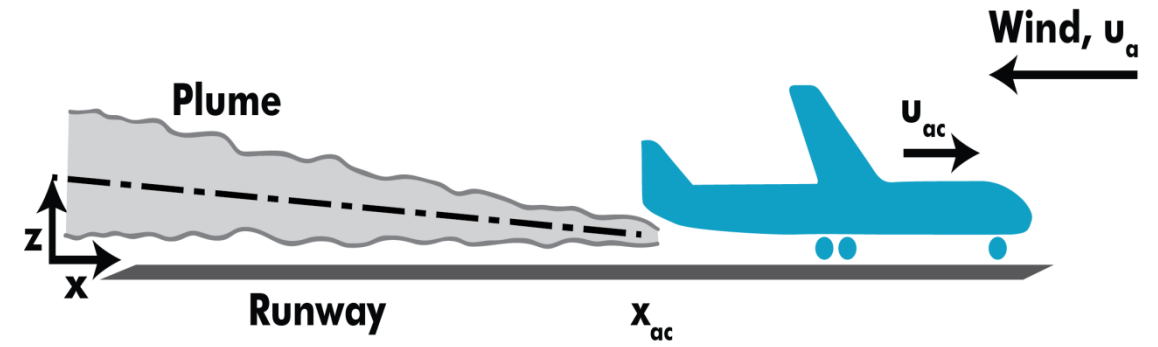


RESULTS



Emissions do not display yet total character and level of aircraft impact on air quality. The factors, which may provide a difference between emission level and air pollution concentrations:

- *type of the engine in aircraft power unit;*
- ***height** of the engine installation at power unit;*
- *the dynamic of exhaust gases jet from aircraft engine;*
- *character of an aircraft movement (parking, taxiing, accelerating on the runway);*
- *meteorological parameters, as wind velocity and atmospheric stability.*

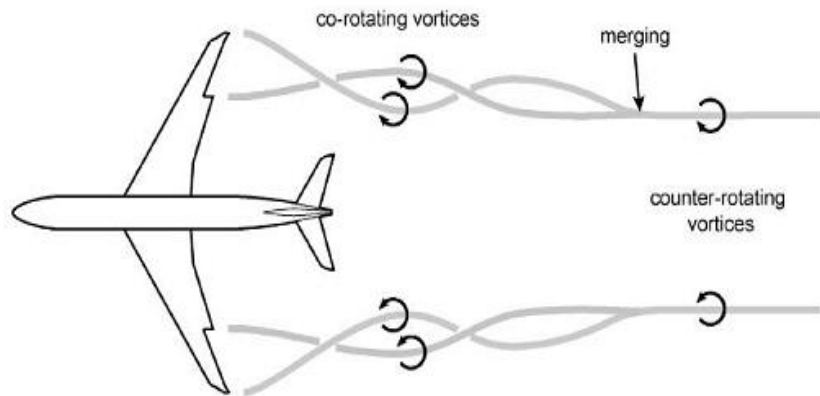




LAQ is about concentration, not emission.

Aircraft is special source of air pollution:

1. **Jet of exhaust gases;**
2. The most part of LTO cycle the aircraft is maneuvering on aerodrome surface, which **significantly impacts on the structure and behavior;**
3. An aircraft wake is composed of **the engine jets**, which are entrained into **the counter-rotating wing (tip, flap) vortices.**



Vortex wake generation behind the aircraft
[F.Garnier, 2005]

The **Gaussian plume model** is used for prediction of vertical and horizontal dispersion of air pollution, it has been adapted for point, line and area sources:

$$C(x; y; z; H) = \frac{Q}{2 \times \pi \times \sigma_y \times \sigma_z \times u} \exp\left[-\frac{1}{2} \left(\frac{y}{\sigma_y}\right)^2\right] \times \\ \times \left\{ \exp\left[-\frac{1}{2} \left(\frac{z-H}{\sigma_z}\right)^2\right] + \exp\left[-\frac{1}{2} \left(\frac{z+H}{\sigma_z}\right)^2\right] \right\}$$

The **initial plume parameters** depend on **aircraft** and **engine type**, **engine operation mode** (jet flow temperature and velocity at exhaust nozzle) and **meteorological conditions** (ambient air temperature, wind velocity and direction).

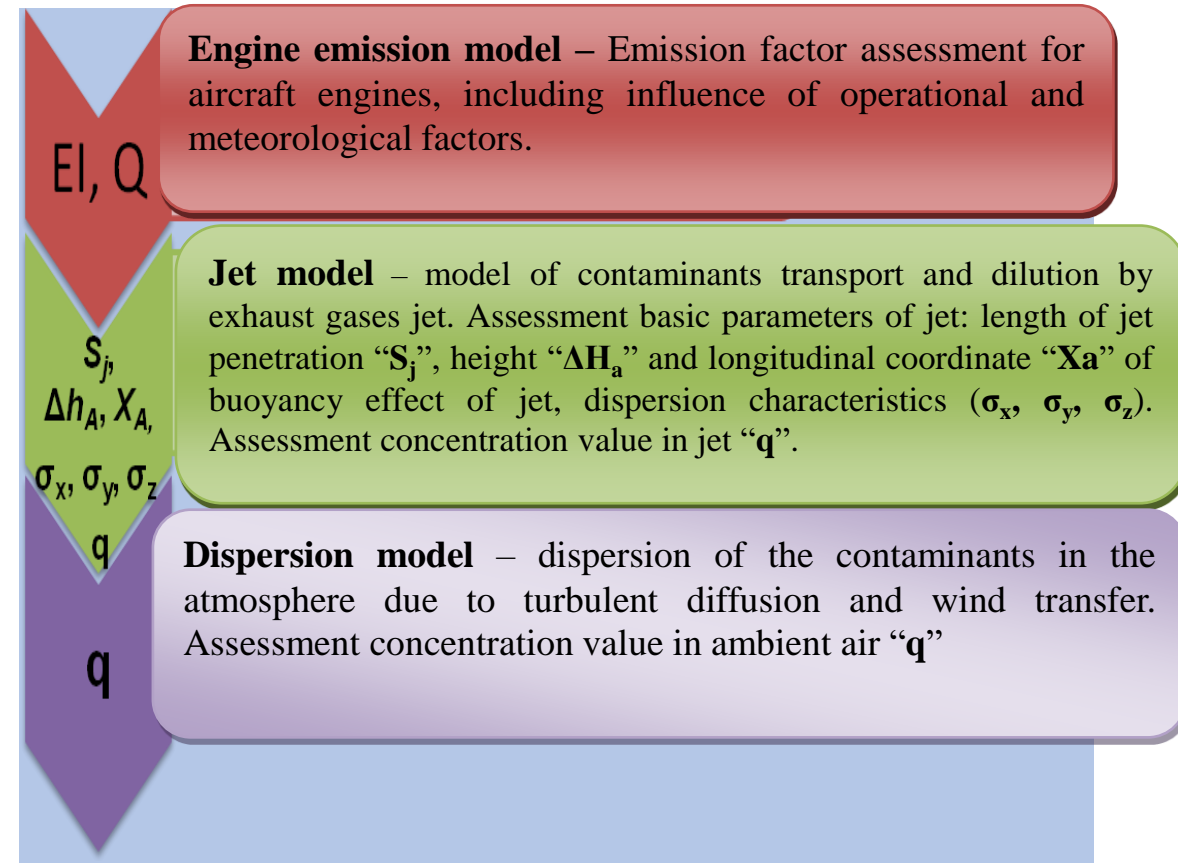


LAQ is about concentration, not emission.



CAEP/12-MDG-FESG/1/WP/13 initiated a feasibility study towards a concentration –based LAQ metric.

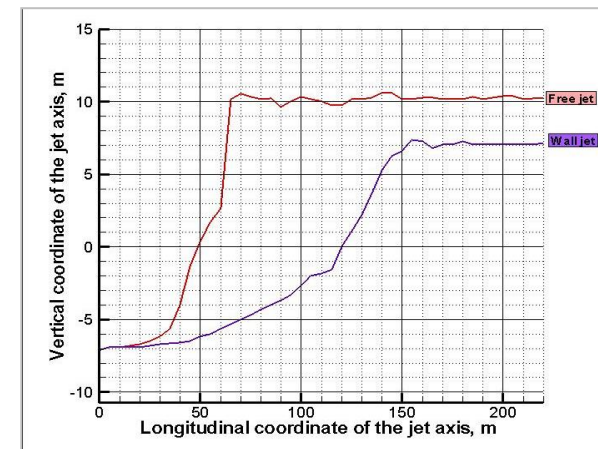
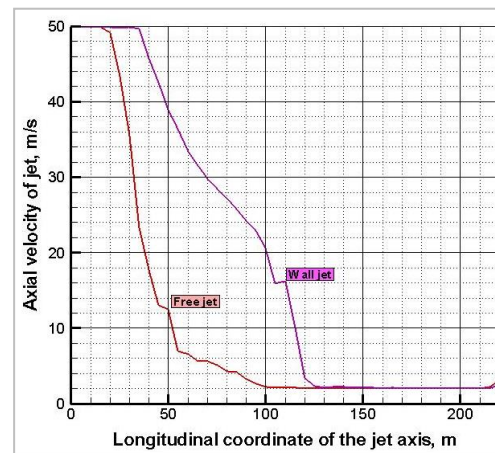
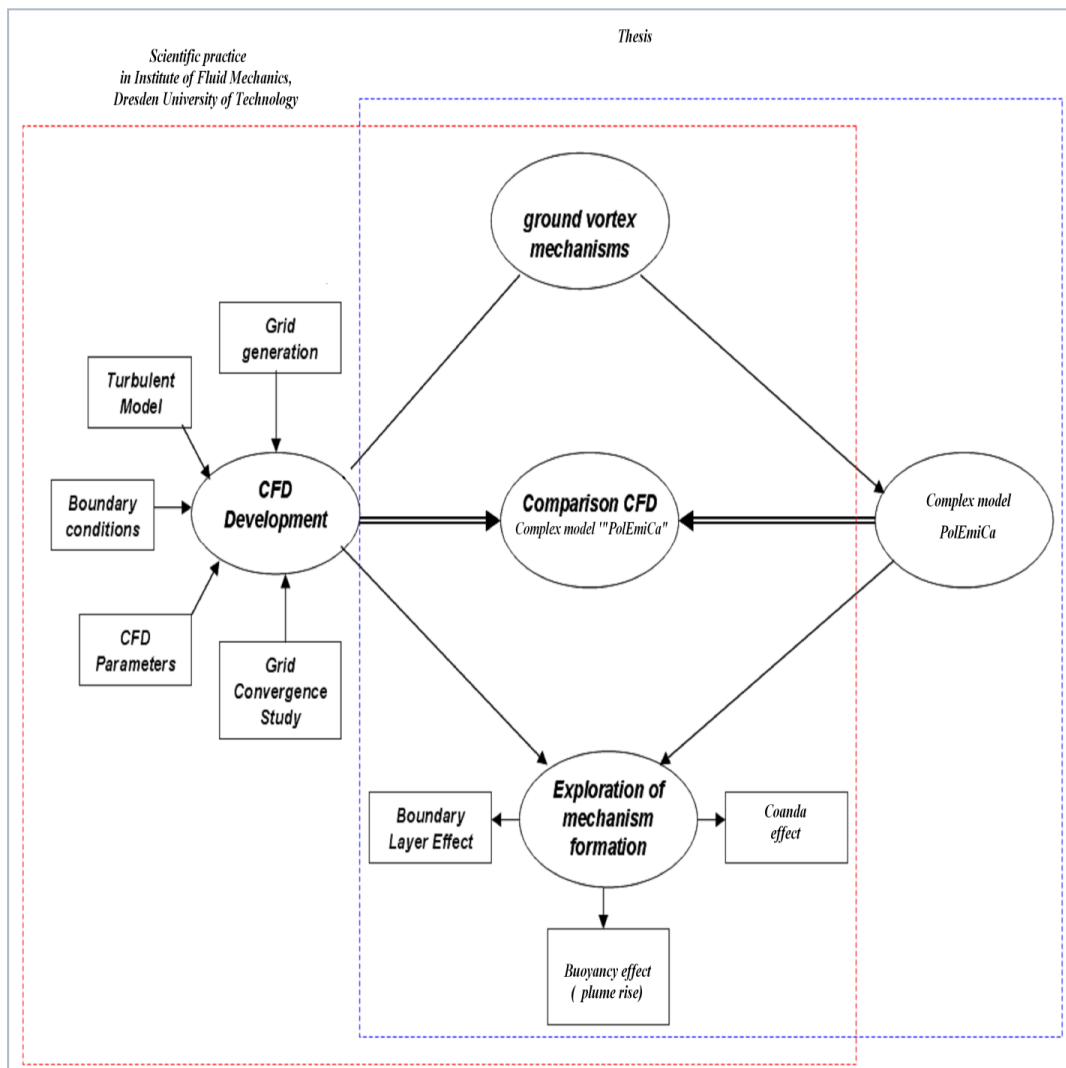
- **LASPORT** is based on **Langragian** dispersion model LASAT (Germany);
- **AEDT** is based on **Gaussian** dispersion model AERMOD (USA);
- **Open-ALAQs** is based on **Gaussian** dispersion model (EUROCONTROL);
- **PolEmiCa** used combined approach of **Eulerian/Gaussian** methods (NAU, Ukraine).



POLEMICA MODEL

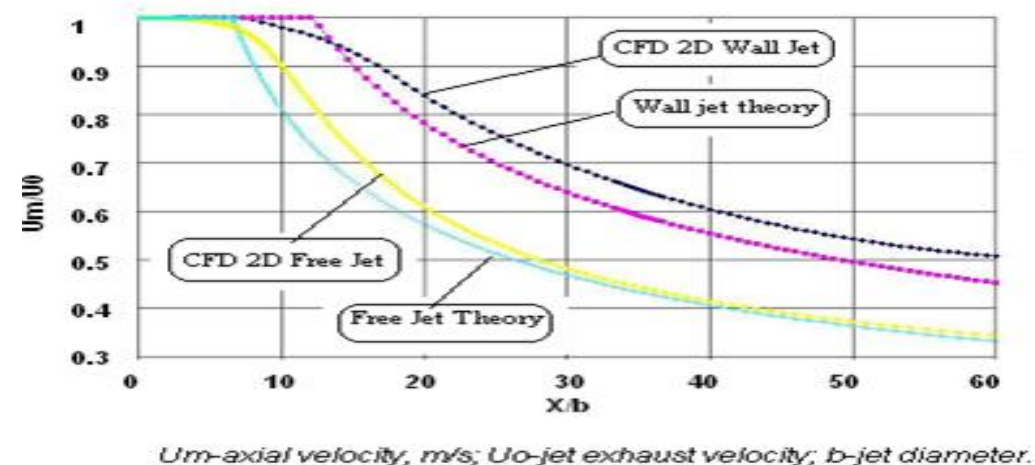
Usually LAQ addresses to emission, usually taken as the integral over the certification LTO, ICAO Doc 9889.

Improvement of complex model PolEmiCa (jet model) by Fluent 6.3



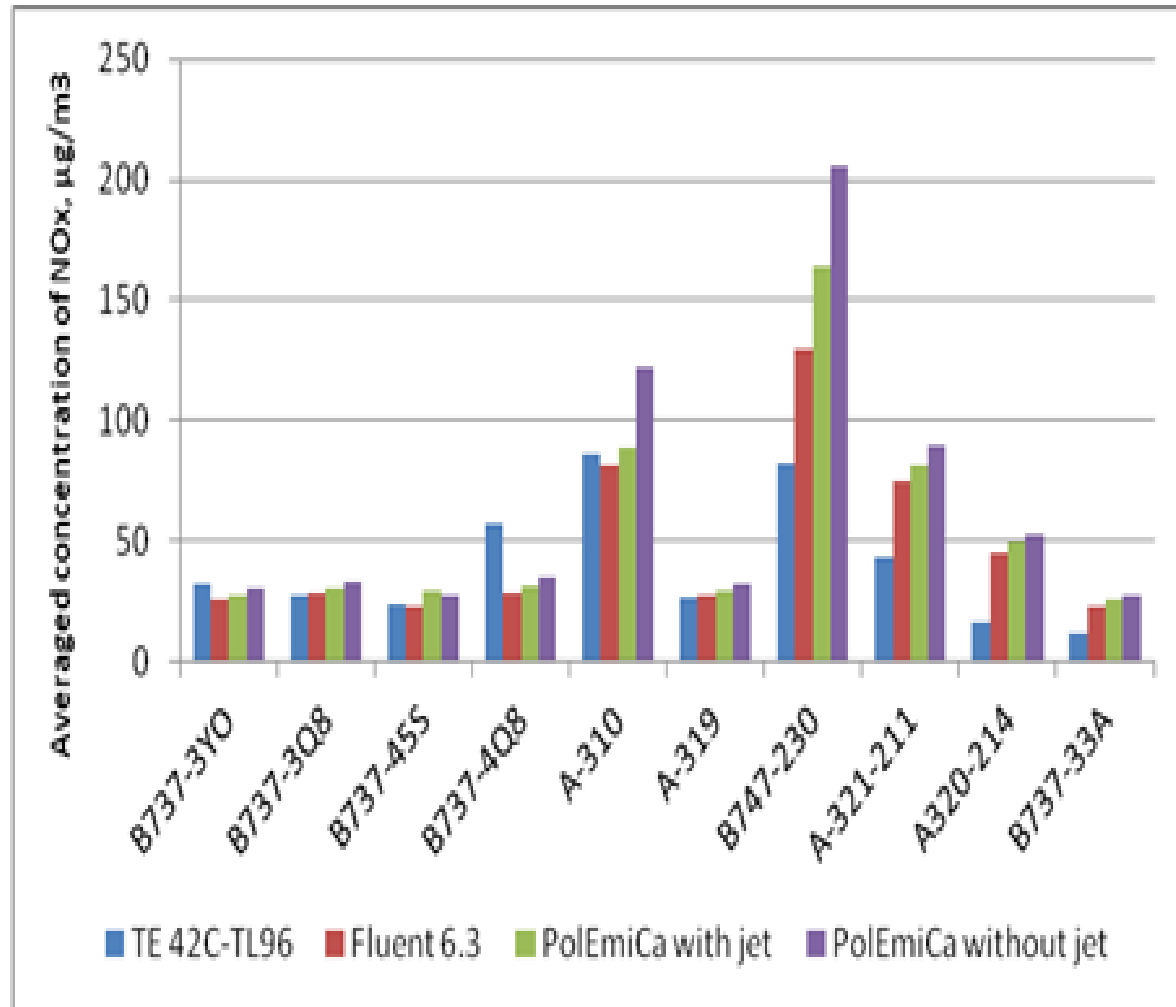
Maximum velocity decay (a) and buoyancy effect (b) of free and wall jet

Obtained by Fluent 6.3 numerical results are supplemented with similar **ALAQS** studies:

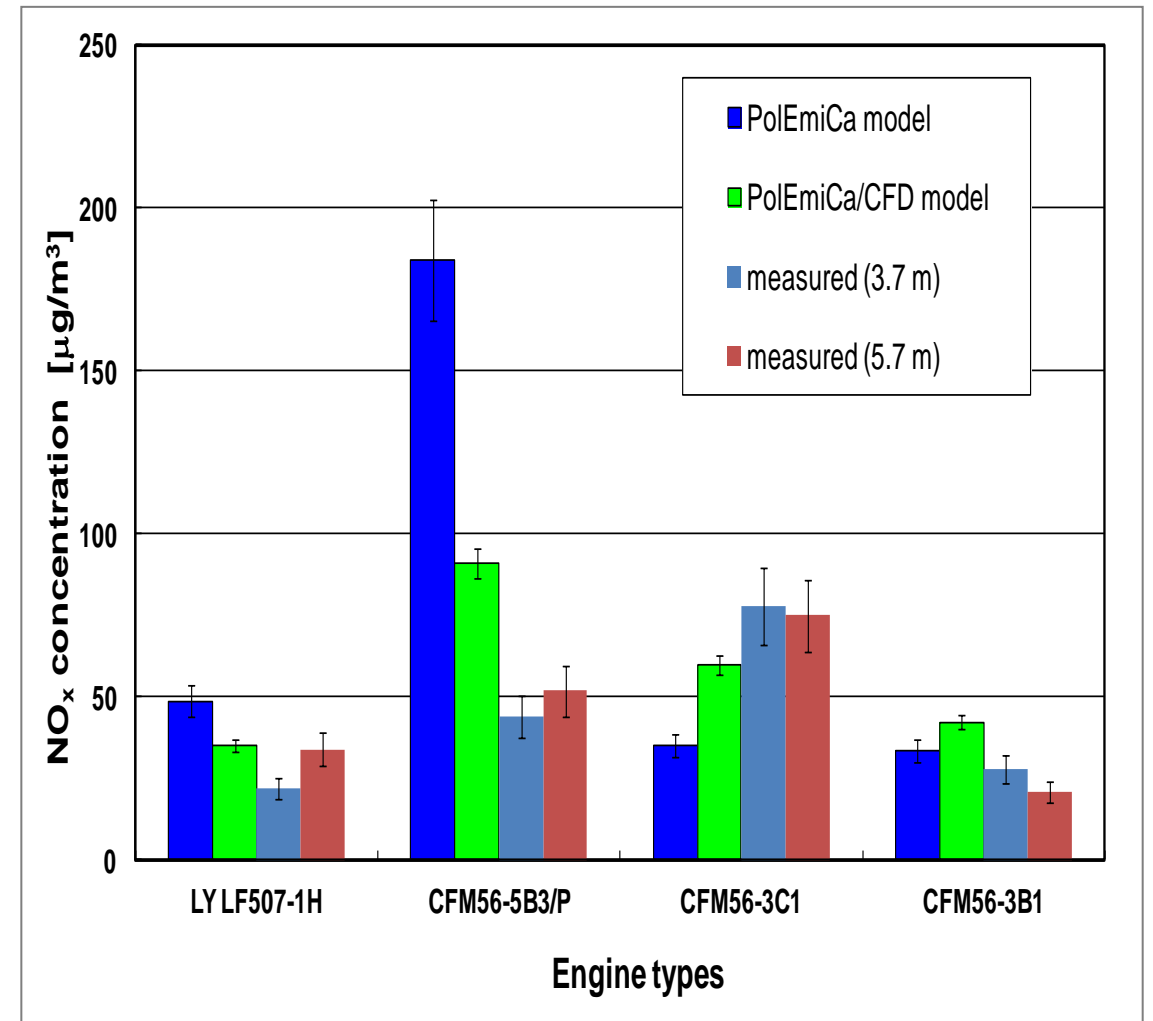


LES method was used to reveal the unsteady ground vortex and turbulence characteristics of fluid flow near aerodrome's surface. **Smagorinsky-Lilly** subgrid-scale model was used to model the smaller eddies (fluctuation component of instantaneous velocity of modeling fluid flow).

Validation of complex model PolEmiCa by measurement campaign at Athens and Boryspil airport

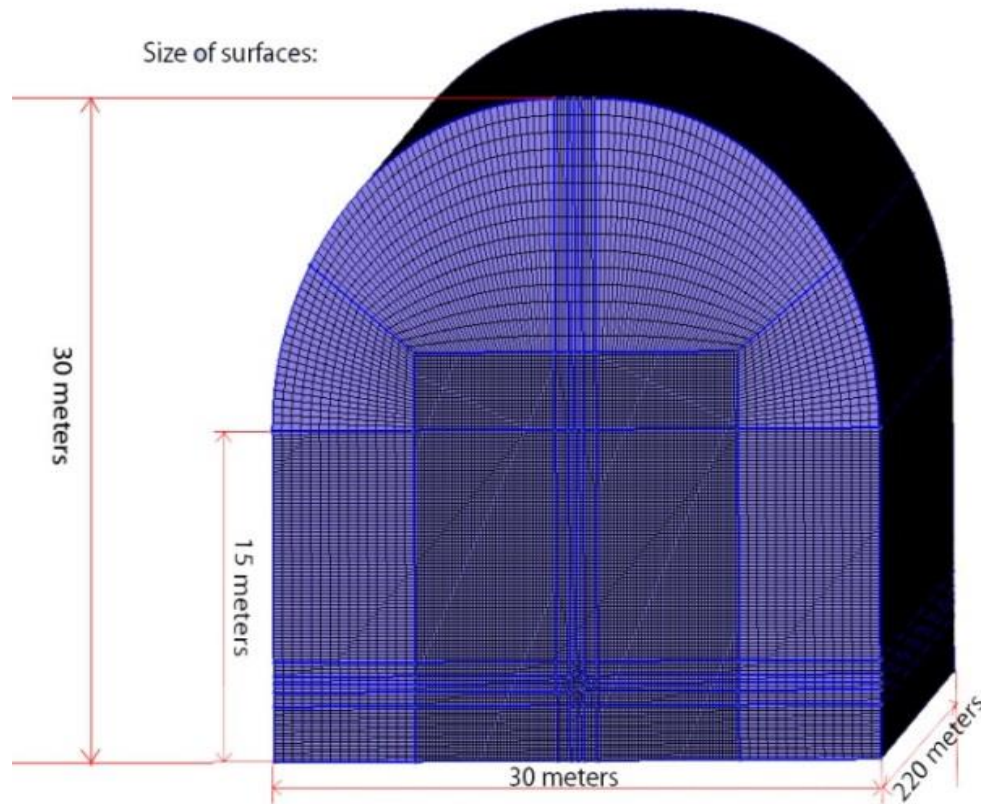


Comparison of measured and modeled averaged concentrations (1 minute) of NO_x in plume from aircraft engine for maximum operation mode



Comparison of measured and modeled averaged concentrations (3 s) of NO_x in plume from aircraft engine for maximum operation mode

OpenFOAM: Computational domain, boundary conditions, turbulence model, solver



Computational mesh in vertical plane

The specific methods:

- **second order** upwind for pressure, momentum, k and ϵ ;
- the **Simple scheme** is used for pressure – velocity coupling.

All calculations were made with steady-state solver (*buoyantBoussinesqSimpleFoam*) for incompressible flows with taking into account temperature difference between the jet ($T=423$ K) and atmospheric air ($T=300$ K).

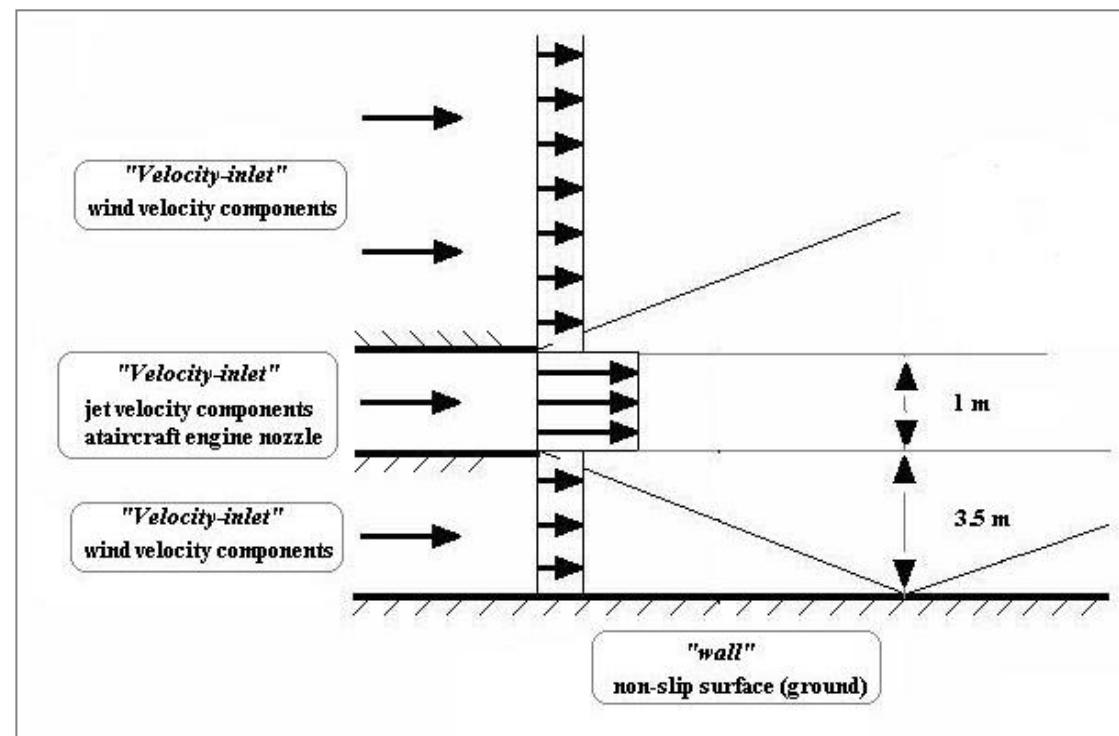
K-epsilon turbulence model was used to evaluate the turbulence characteristics of fluid flow.



Boundary conditions



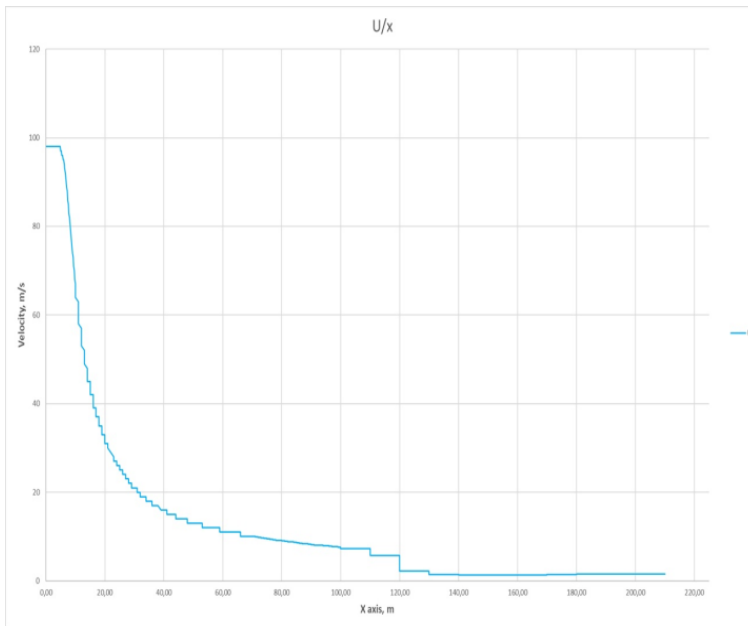
- The nozzle section of aircraft engine exhaust is set as a “**velocity inlet**” with velocity magnitude 98 m/s and temperature 423 K ;
- The computational surfaces adjacent to the engine section at which ambient conditions is also set as “**velocity inlet**” with wind velocity $2 \text{ m} \cdot \text{s}^{-1}$ and temperature 300 K;
- The external lateral surfaces of computational domain at which ambient conditions are set also as “**velocity inlet**”;
- The ground surface, which is corresponding to the bottom of the computational domain, is set as “**wall**” implying a non-slip condition for velocity and with temperature 300 K.
- The computational surface opposite to the aircraft engine exhaust nozzle, at which flow field (mixture jet and ambient air) leaves computational volume, was set as “**pressure-outlet**”.



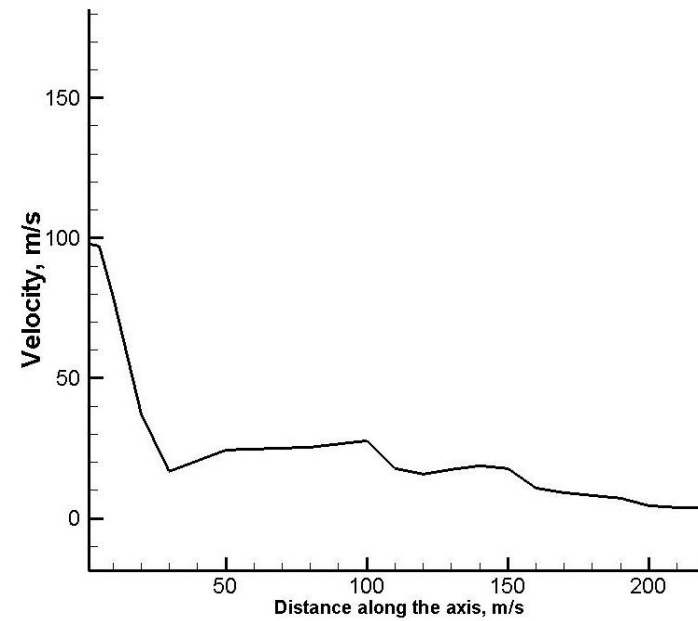
Boundary conditions for CFD simulations of exhaust gases jet from aircraft engine near ground



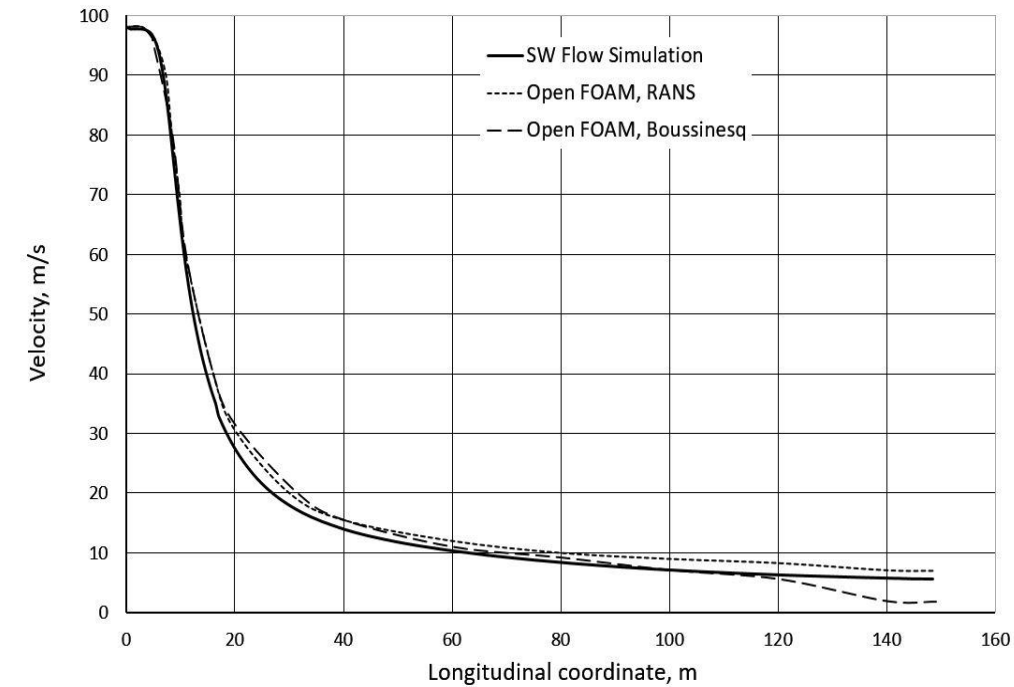
Results of simulation



a)



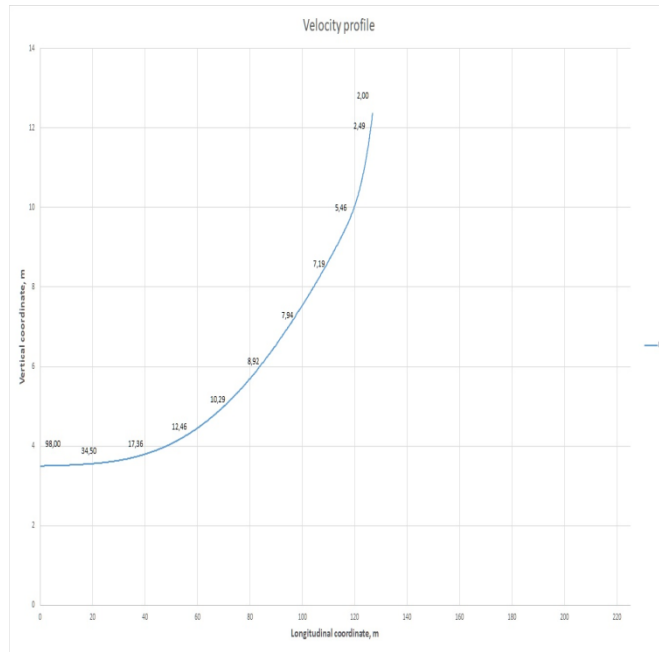
b)



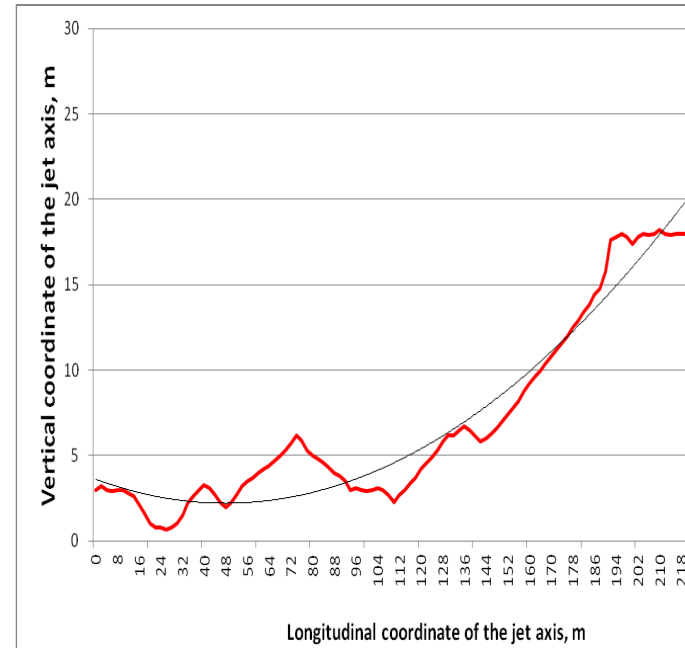
c)

Maximum velocity decay along the axis of the jet due to OpenFOAM (a) Fluent 6.3 (b), Flow Simulation (c)

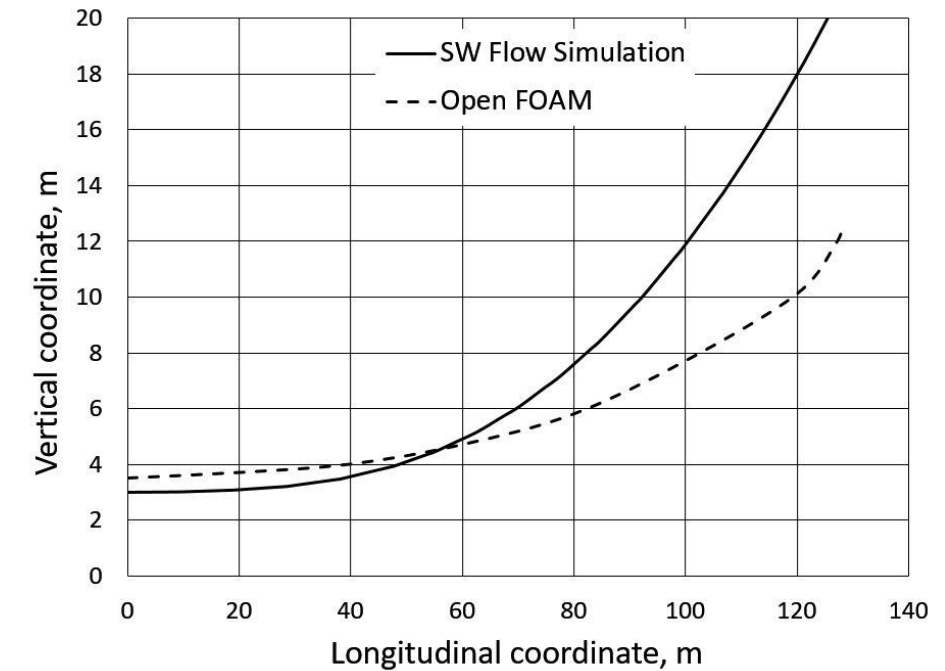
Results of simulation



a)



b)



c)

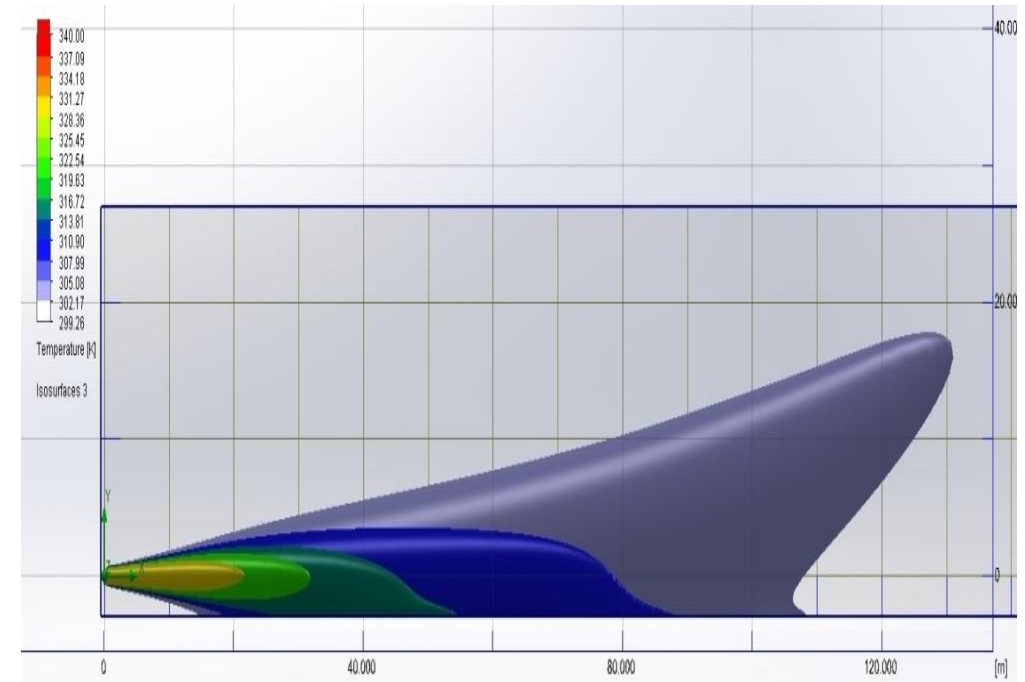
Buoyancy effect of exhaust gases jet:
longitudinal and vertical coordinates of jet axis due to OpenFOAM (a), Fluent 6.3 (b) and Flow Simulation (c)



Results of simulation:



Parameters	Calculation results of modeling			
Initial conditions	Open FOAM	Fluent 6.3	Fluent 6.3	PolEmiCa
$U_j = 98 \text{ m}\cdot\text{s}^{-1}$, $T_j = 423\text{K}$		Mesh 1	Mesh 2	previous version
Height of jet rise Δh_A , m	8,5	15,5	17,0	40,4
Longitudinal coordinate of jet rise X_A , m	130,0	170,0	165,0	125,0
Length of jet penetration, S_{jet}	190	205	210	200



Comparison between OpenFOAM numerical results and semi-empirical jet model calculations (used by complex model PolEmiCa) show that buoyancy effect parameter as height decreases in 4 times wall jet.

CONCLUSIONS:



The **specific features of aircraft** as a special source of air pollution, needs to be included **in emission and dispersion modeling** of airport air quality. Particularly, eliminating the fluid dynamics of the jet from aircraft engines may overestimate the height of the buoyancy of the exhaust gases jet, underestimate its length and radius of expansion, dispersion characteristics and accordingly pollutant concentrations.



Thank you for your attention!

